# Zelf Vliegen

Checklist
Aero AT-3





This checklist is in accordance with the Aero AT-3 R100 Aircraft Flight Manual section 3 and 4, revision 14, November 2012

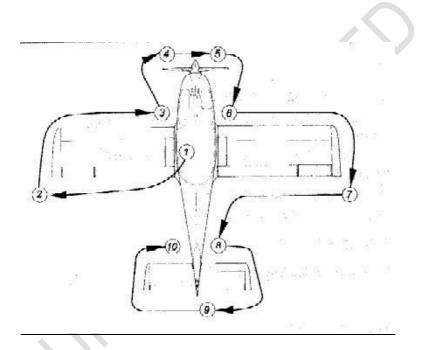
Aero AT-3 PH-EVB Checklist

ATTENTION!
DO NOT
STOW THIS CHECKLIST
IN DIRECT SUNLIGHT

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# **Preflight Inspection Aero AT-3 PH-EVB**

It is the duty of the pilot to perform a pre-flight inspection prior to the flight or after a break in flights, when he has left the cabin. The inspection is to be made, starting with the cabin and walking clockwise around the aeroplane.



1.CABIN	
Canopy – Opening, closing and	CHECK
operation of locks	
Inside cabin – All foreign items	REMOVE
Collapsible tow bar from flying controls – if installed	REMOVE
Condition of the seats	CHECK
Luggage collapsible tow bar – if will carry	SECURE
Seat belts	CHECK
Flight controls – Free movement,	CHECK
lack of significant play and extensive friction	
Balancing tab (trim)	CHECK
- Full travel	ESTABLISH
- Take-off setting	
Wing flap – Extension	CHECK
Wing flap setting to 40°	ESTABLISH
Carburettor heating – to be set OFF	CHECK
Fuel valve – to be set OFF	CHECK
Fuel pump – to be switched OFF	CHECK
Ignition – to be set OFF	CHECK
Fuel level – to be checked with the	CHECK
gauge	
Battery and generator – to be	CHECK
switched OFF	
All electrical equipment – to be switched OFF	CHECK
Circuit breakers	CHECK IN
Parking Brake (if installed)	ON

2.PORT WING	
Structure – Condition and	CHECK
cleanliness	
Wing flap – Condition of structure and play in control system and	CHECK
hinges	
Ailerons – Condition of structure and	CHECK
play in control system and hinges	
Pitot tube – Fixing and cleanliness	CHECK
Inspection flap – to be closed and	CHECK
locked	

3.PORT LANDING GEAR		
Tire – Check the tyre pressure	CHECK	
(visually)		
Brake system	CHECK	

4.FUSELAGE FRON	T PART
Canopy – Visually check cleanliness	CHECK
Fuel tank – Fuel quantity and locking	CHECK
the filler-cap	
Engine cowling – Locking and leaks	CHECK
Propeller and spinner – Condition	~
and cleanliness	CHECK
Exhaust pipes – Condition	CHECK
Antenna of transponder – Condition	CHECK
and fixing	
Fuselage bottom surface – Condition	CHECK
and cleanliness	
Air intake covers – installed	CHECK

## NOTE:

It is recommended that Air Intake Covers are installed when operating the aircraft in ambient temperature below 12 °C

5.NOSE LANDING GEAR	
Tire – Check the tyre pressure	CHECK
(visually)	
Shock absorber – Condition	CHECK
Towing bar – to be removed from	CHECK
the aeroplane	

6.STARBOARD LANDING GEAR AND FRONT PART OF FUSELAGE	
Tyre – Check the tire pressure (visually)	CHECK
Brake system	CHECK
Oil level and presence of the dipstick (turn the propeller several times first	CHECK
by hand)	

#### **CAUTION:**

When turning the propeller by hand, special care is to be observed and the following is to be checked:

- -the ignition is switched off
- -the parking valve is on, or
- the chocks are put under wheels

The possibility of spontaneous ignition always exists

7.STARBOARD WING	
Structure – Condition and	CHECK
cleanliness	
Ailerons – Condition of structure and	CHECK
play in control system and hinges	
Wing flap – Condition of structure	CHECK
and play in control system and	
hinges	
Inspection flap – to be closed and	CHECK
locked	

8.FUSELAGE REAR PART, STARBOARD	
Structure – Condition and cleanliness	CHECK
Antennae – Condition and cleanliness	CHECK

9.EMPENNAGE	
Fin – Condition and cleanliness	CHECK
Rudder – Hinges and their play	CHECK
Slab tail – Hinges and their play	CHECK
Trim & balancing tab – Hinges and	CHECK
their play	

10.FUSELAGE REAR PART, PORT	
Structure – Condition and	CHECK
cleanliness	
Inspection flap – to be locked	CHECK

# Normal checklist Aero AT-3 PH-EVB

BEFORE STARTING	G ENGINE
Canopy	SHUT AND LOCK
Luggage – stow and secure	CHECK
Seat belts	FASTEN
Reading of the fuel quantity indicator	CHECK
Ignition to be switched off	CHECK
Battery, generator and alternator  – to be switched off	CHECK
All electrical equipment – to be switched off	CHECK
Trim and balancing tab – to be set to "TAKEOFF"	CHECK
Flight controls – full and free movement	CHECK
Wing flaps	RETRACT
ELT	ARMED

#### **CAUTION:**

Engine started is prohibited with the parking brake on

ENGINE START		
Fuel valve – to set	OPEN (push forward)	
Choke (if cool engine)	ON (lock fully out)	
Battery , generator and alternator	ON	
"GENERATOR / ALTERNATOR	CHECK	
FAILURE" light - illumination		
Fuel Pump	ON	
Throttle lever	IDLE (or open by 10%)	
Anti collision Light	ON	
Propeller area	CLEAR	
Parking Brake	OFF	
Brakes	APPLY	
Ignition switch	START	

#### **NOTE**

After completing the engine start, check whether the oil pressure starts to rise within 10 sec. (above 2 bar)

AFTER ENGINE	START
Time	NOTE
Engine Speed 2500 RPM	MAINTAIN
GEN/ALT FAILURE light – go out	CHECK
STARTER ENGAGED light – go out	CHECK
Choke	OFF
Oil heater	ON (as required)
Oil pressure – green arc	CHECK
Fuel pump	OFF
Electrical equipment	ON (except landing -, wing
	tip strobe, position light)
Flight Instruments	CHECK
Engine instruments	CHECK
Engine speed of 2000-2500 RPM –	MAINTAIN
until oil temperature of 50°C is achieved	
Avionics Master Switch	ON
Aspen PFD	ON
PCAS	ON
Position light	ON (as required)
Taxi light	ON (as required)
Altimeter	SET, 2 x
Radio	SET ON AND CHECK
Transponder	SET, STBY (standby)
Circuitbreakers	CHECK ALL IN

TAXI	
Brakes	RELEASE
Operation of the brakes	CHECK
Gyros & Compass	CHECK in turn L/R
Aspen PFD	CHECK in turn L/R
Control Stick	According to wind

## **CAUTION:**

To avoid engine overheating and pollution with dust operation of the engine on ground at ratings higher than the required for taxiing is to be limited to a minimum

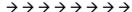
ENGINE TEST	RUN
Brakes	APPLY
Control stick	PULL
Indications of engine instruments – to be within the green sector of the scale	CHECK
Engine speed 4000 RPM	SET
Ignition switch	CHECK R/L (drop max 300)
Ignition switch	вотн
Carburettor heating	CHECK (10 sec)
Engine instruments	CHECK
Warning lights	CHECK
Throttle	IDLE (~1600 RPM)
Throttle	2000 RPM

#### NOTE:

Maximum engine speed on ground is 5050 RPM RPM drop when one ignition unit only operating is 300 RPM. Maximum difference of engine speed between position "L" and position "R" must not exceed 120 RPM

BEFORE TAKE-OFF	
Fastening of the seat belts	CHECK
Fuel valve - Open	CHECK
Canopy (window)	LOCKED (2x)
Trim and balancing tab – to be set	CHECK
for take-off	
Wing flaps – to be set for take-off	CHECK
(or grass: 15 degrees)	
Ignition switch – to be set BOTH	CHECK
Carburettor heating	OFF

## BEFORE TAKE-OFF CHECKLIST TO BE CONTINUED NEXT PAGE!!!



BEFORE TAKE-OFF (CO	NTINUATION)
Oil pressure – to be in green sector	CHECK
Oil temperature – to be in green	CHECK (min 50°C)
sector	
Cylinder heat temperature	CHECK
EGT temperature	CHECK
Fuel pressure – to be in green	CHECK
sector	
Amp. / Volt meter	CHECK
Altimeter – to be set properly	CHECK
Landing light	ON
Wingtip strobe light	ON
Fuel pump	ON
Transponder	ALT
Windsock	CHECK

# ONLY FOR RFI AND FI "GOPRO" SWITCH ON

TAKE-OFF	
Brakes	RELEASE
Throttle – full open	SET
Maximum engine speed	CHECK (max. 5050 RPM)
Take-off direction –maintain using	EXECUTE
rudder pedals	
Airspeed after lift-off to be	60 kts
maintained at	
Landing gear – rotating wheels	BRAKE
When height 50 feet reached -	65 kts
Increase to speed to	
Wing flaps	RETRACT

AFTER TAKE-OFF	
Engine instruments	CHECK
Fuel pump	OFF
(On top of climb or above 1000 ft)	
Flaps	RETRACT
Landing light	OFF

CLIMB	
Throttle – to be opened	5000 RPM
Airspeed – for climb, to be	65 kts
maintained	
Engine operational parameters –	MONITORED
to be	

CRUISE	
Throttle – as required	SET
Trim and balancing tab – as for	SET
cruise	
Engine operational parameters –	MONITORED
to be	
Oil heater	ON (check temperature)
	as required

DESCENT	
Throttle – as required	SET
Fuel pump	ON
Coolant an oil temperature – to be	MONITORED

BEFORE LANDING	
Warning Lights	CHECK
Engine instruments	CHECK
Fuel pump	ON
Carburettor heating – as required	SET
Flaps	SET
Airspeed	SEE TABLE "OPERATING SPEEDS"
Landing light	ON

LANDING	
Engine RPM below 50 feet to be	DIMINISHED
Touch down with the main wheels first	
Throttle	IDLE
Braking	AS REQUIRED

BALKED LANDING (GO-AROUND)		
Carburettor heating	OFF	
Throttle – gradually	FULL OPEN	
Airspeed – to be	INCREASED	
Wing flaps – gradually	RETRACT	
Airpseed – to be maintained:	65 kts	
Proceed to climb	EXECUTE	

AFTER LANDING		
Flaps	RETRACT	
Carburettor heating	OFF (recheck)	
Wingtip strobe light	OFF	
Landing light	OFF	
Positon light	OFF	
Turn Coordinator	OFF	
Aspen PFD	OFF	
Fuel Pump	OFF	
Transponder	Set STBY (standby)	
PCAS	OFF	

ENGINE SHUTDOWN	
Radio	OFF
Taxi light	OFF
Transponder	OFF
Electrical Equipment	OFF
Avionics Master	OFF
Throttle – to be set to	IDLE
Ignition Switch	OFF
Keys	REMOVE
Battery ,generator and alternator	OFF
Anti collision light	SWITCHED ON at all times
Time	NOTE

OPERATING SPEEDS		
Take-off	42 KIAS to 50 feet	
	60 KIAS above 50 feet	
Take-off degrees 0	50 KIAS	
Beste Angle (Vx) degrees 0	60 KIAS	
Beste Rate (Vy) degrees 0	65 KIAS	
Best Angle (Vx) degrees 15	55 KIAS	
Best Rate (Vy) degrees 15	60 KIAS	
Max Turbulent	85 KIAS	
Glide	65 KIAS	
Cruise	90 KIAS 4600 RPM	
Downwind flaps 15 degrees	80 KIAS 4200 RPM	
Base flaps 15 degrees	70 KIAS 3200 RPM	
Final 15 degrees	65 KIAS 3200 RPM	
Final 40 degrees	60 KIAS	
Crosswind max	11,7 kts	

# **Emergency checklist Aero AT-3 PH-EVB**

#### **ENGINE FAILURES**

ENGINE FAILURE AFTER TAKE-OFF (EFATO)		
Maintain airspeed	65 kts	
Fuel pump	OFF	
Fuel valve	SHUT	
Throttle	IDLE	
Ignition Switch	OFF	
Battery, generator and alternator	OFF	
Landing: ahead avoiding obstacles		

ENGINE FAILURE II	VELIGHT	
FUEL PRESSURE DROP ENGINE POWER DROP		
Fuel pump	ON	
Carburettor heating	Switch ON	
Fuel valve opening	To be CHECKED	
Fuel quantity on board	To be CHECKED	
EXCESSIVE ENGINE VIBRATION		
Carburettor heating	Switch ON	
Fuel pump	Switch ON	
EXCEEDING THE CYLINDER HEAD TEMPERATURE		
Temp of the exhaust gases for	To be CHECKED	
comparison		
Over-speeding the engine		
Exceeding the max oil temperature		
The oil Temperature drops below		
the permissible minimum		

#### **CAUTION:**

In all of the above cases reduce the power to the minimum possible fly to the nearest airfield and be prepared for precautionary landing

ENGINE RESTARTING	IN FLIGHT	
Maintain airspeed	65 kts	
Fuel quantity in tank	To be CHECKED	
Fuel valve	OPEN	
Fuel pump	Switch ON	
Throttle to be set	IDLE (or 10% opening)	
Choke (when the engine is cool)	ON	
If the prop windmill – ignition	ON	
If the prop stopped – engine starter	ON	
If the engine starts		
Throttle, according to the required	SET	
power		
Operational parameters of the	CHECKED	
engine		
Emergency fuel pump	OFF	
If the engine does not start		
Perform EMERGENCY LANDING		

#### NOTE:

The engine can be re-started in the entire range of operational airspeeds and altitudes. The loss of altitude and airspeed during engine re-starting in flight is not great. No other special procedures are required for engine re-starting in flight.

#### **SMOKE AND FIRE**

ENGINE FIRE ON GROUND	
Fuel Valve	SHUT
Throttle	FULL OPEN
Ignition Switch	OFF
Electrical Equipment	OFF
Battery, generator and alternator	OFF
Fire extinguisher	To be USED

FIRE IN FLIGHT	
Maintain airspeed	65 kts
Fuel Valve	PULL SHUT
Throttle	FULL OPEN
Ignition Switch	OFF
Battery, generator and alternator	OFF
Cabin canopy vents	SHUT
A side slip – opposite fire to blow it	TO BE PERFORMED
out	
When the engine stops	PERFORM EMERGENCY
	LANDING

# **CAUTION:** After an engine fire do not try to re-start the engine

FIRE IN THE ELECTRICAL SYSTEM	
Maintain airspeed	65 kts
Electrical equipment	OFF
Fire extinguisher (if fire in cabin)	To be USED
Cabin canopy vents	KEEP OPEN
If the fire persists, decide upon a place for landing	

GLIDING FLIGHT	
Recommended aeroplane	Flaps retracted
configuration	
Airspeed	65 kts
Throttle	IDLE
Gliding ratio (no power)	8

#### **EMERGENCY LANDING**

PRECAUTIONARY LANDING		
Landing place	IDENTIFY	
Wing flaps to 40 degrees	EXTEND	
Maintain approach airspeed	54 kts	
Safety belts	FASTEN FIRMLY	
Electrical equipment	OFF	
Locks of the canopy	UNLOCK	
BEFORE TOUCH DOWN		
Fuel valve	PULL SHUT	
Battery, generator and alternator	OFF	
Ignition switch	OFF	
Levelling out directly before touch-down. After touching down		
keep control stick fully pulled.		

NE FAILURE
EXTEND
54 kts
FASTEN FIRMLY
UNLOCK
OFF
PULL SHUT
OFF
OFF
IDLE

RECOVERING FROM UNINTENTIONAL SPIN		
Throttle	IDLE	
Rudder opposite to aeroplane rotation	APPLY	
Control stick	NEUTRAL	
Ailerons	NEUTRAL	
Wing flaps	RETRACT	
WHEN THE AEROPLANE STOPS TO ROTATE		
Rudder	NEUTRAL	
Control stick – gentle proceed to level flight		
Throttle for level flight	To be set	

# WARNING Intentional spinning is prohibited

## **OTHER EMERGENCY PROCEDURES**

ICING		
The aeroplane is not equipped with a de-icing system. Therefore		
the area where icing conditions exist is to be left as soon as		
possible.		
Carburettor heating	ON	
Heating of the cabin	ON	
To a limited degree some ice may be removed by hand, through		
the window of the cabin.		

ABANDONING THE AEROPLANE W	TH USE OF PARACHUTE
Maintain airspeed	65 kts
Fuel valve	PULL SHUT
Ignition switch	OFF
Battery and generator	OFF
Headset cables	DISCONNECT
Safety belts	UNFASTEN
Canopy (Pull both jettisoning levers and push out the canopy both hands)	TO BE JETTISIONED
The aeroplane	TO BE ABANDONED
The parachute at a safe distance	DEPLOY

FAILURE OF THE ELECTRIC SYSTEM		
Check the condition of the system		
(voltammeter generator signalling		
light)		
Check the circuit breakers and fuses.	Switch ON again, as	
	required	
In case of generator failure act as follows		
Generator	OFF	
Power receivers not required to	OFF	
continue the flight		

#### FAILURE OF THE STATIC AND PITOT PRESSURE SYSTEMS

The failure of the flight and navigation instruments might be caused by leakage or constriction of the pipes of the static or pitot pressure systems.

In case of failure of the static pitot pressure system, the landing approach is to be performed with flight parameters monitored by the tachometer and other correctly working flight and navigational instruments only. On ground, water sediment is to be removed from the systems, and the sensors of static and pitot pressure checked to be clean and not constricted. Have the systems checked for leakage.

#### FAILURE OF BALANCING TAB CONTROL SYSTEM OF SLAB TAIL

In case of failure of the balancing tab control system of the slab tail in flight, if the aeroplane becomes "tail heavy" (the nose rises), the airspeed is to be reduced to read about IAS 60 kts to reduce the force on the control stick.



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